

IN THE SPECIFICATION

Please replace the paragraph beginning at page 12, line 15, with the following rewritten paragraph:

In the gas turbine of the sixth embodiment, the curvature from the back side 6 throat SO to the rear edge 6 of the final stage moving blade 4S is defined in a ~~rang~~range of 0 to 0.15. As a result, the gauging ratio of the final stage stationary blade 4C of the gas turbine in the second embodiment and the exit angle  $\alpha_2$  ratio of the final stage stationary blade 4C of the gas turbine in the third embodiment can be determined appropriately.